

Concept of a Lunar ISRU Plant

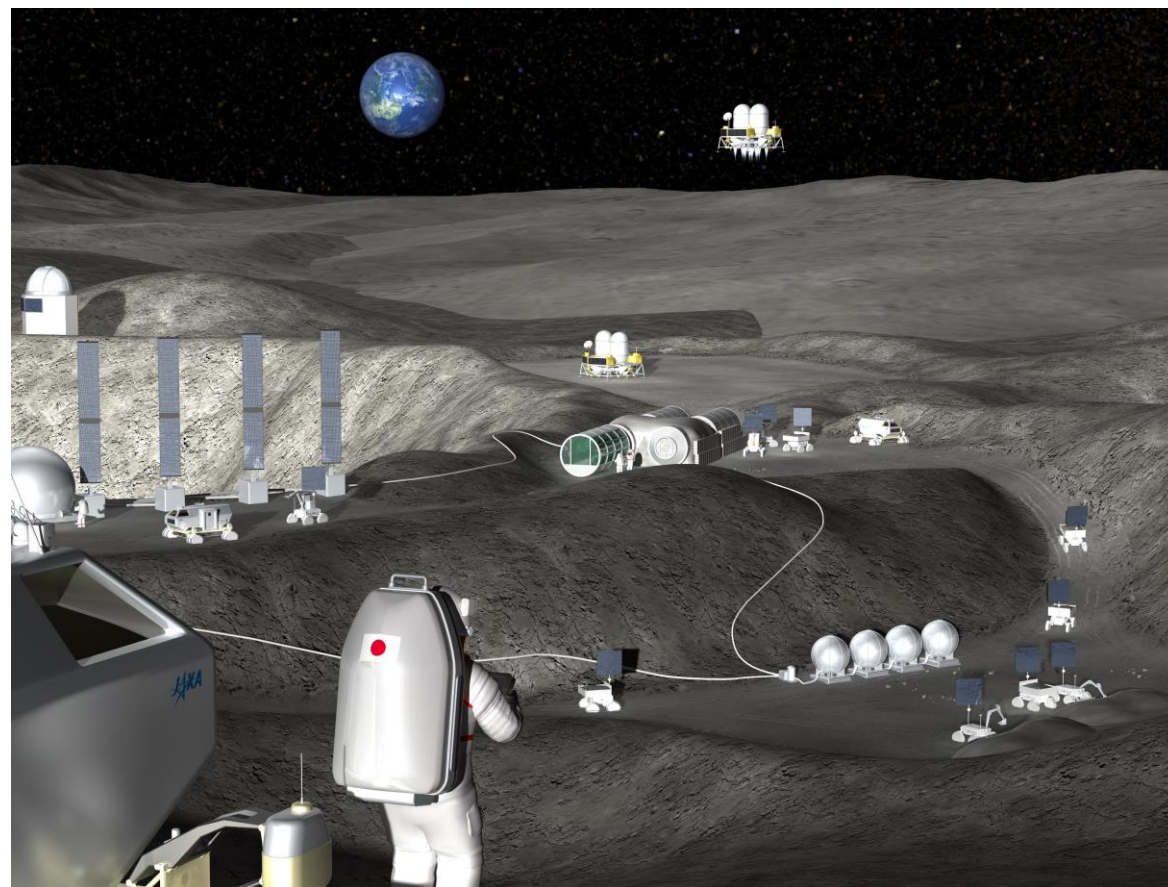
- In-situ Production of LOX / LH2 from Lunar Regolith -

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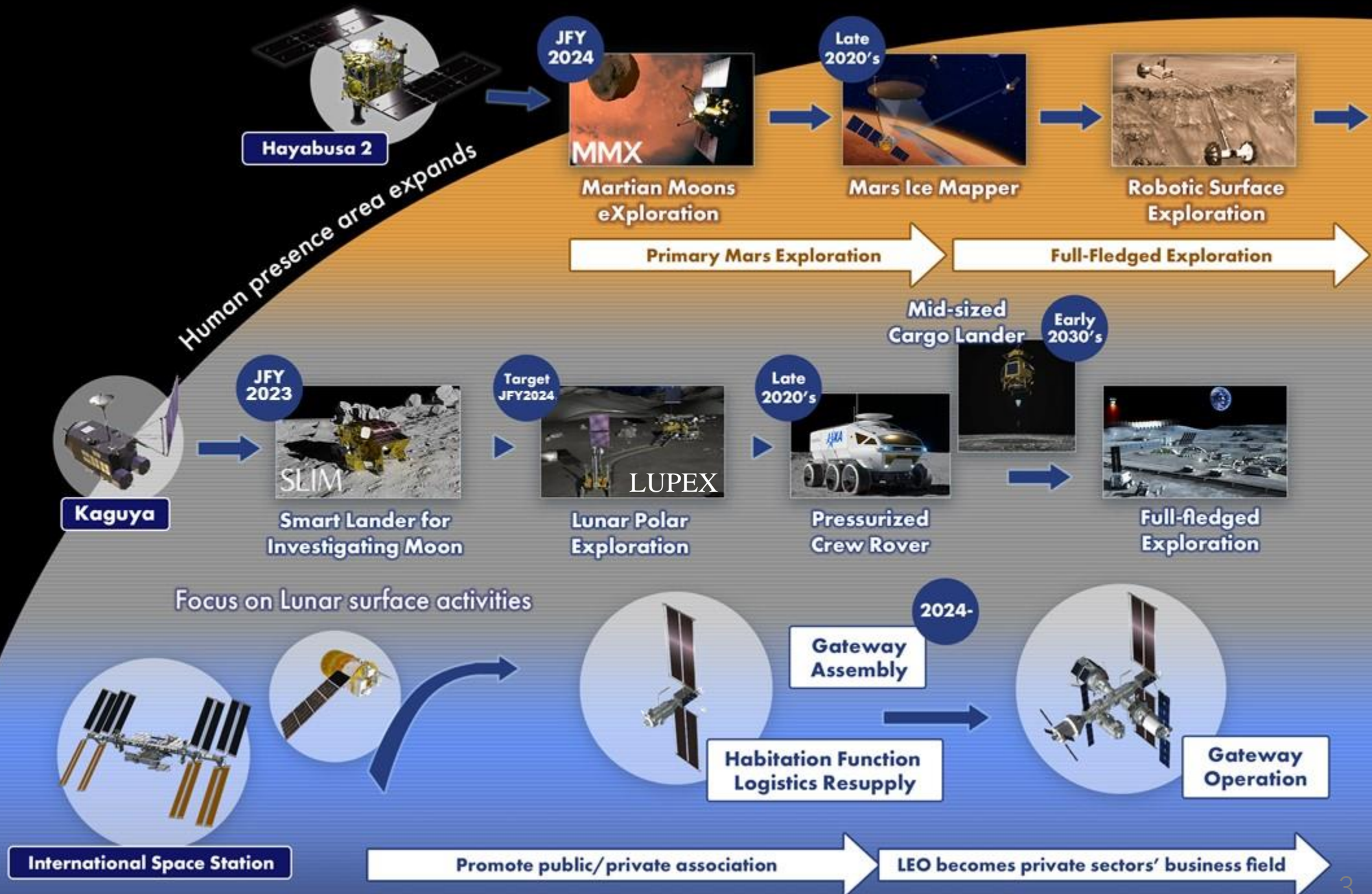
MARS



MOON



EARTH



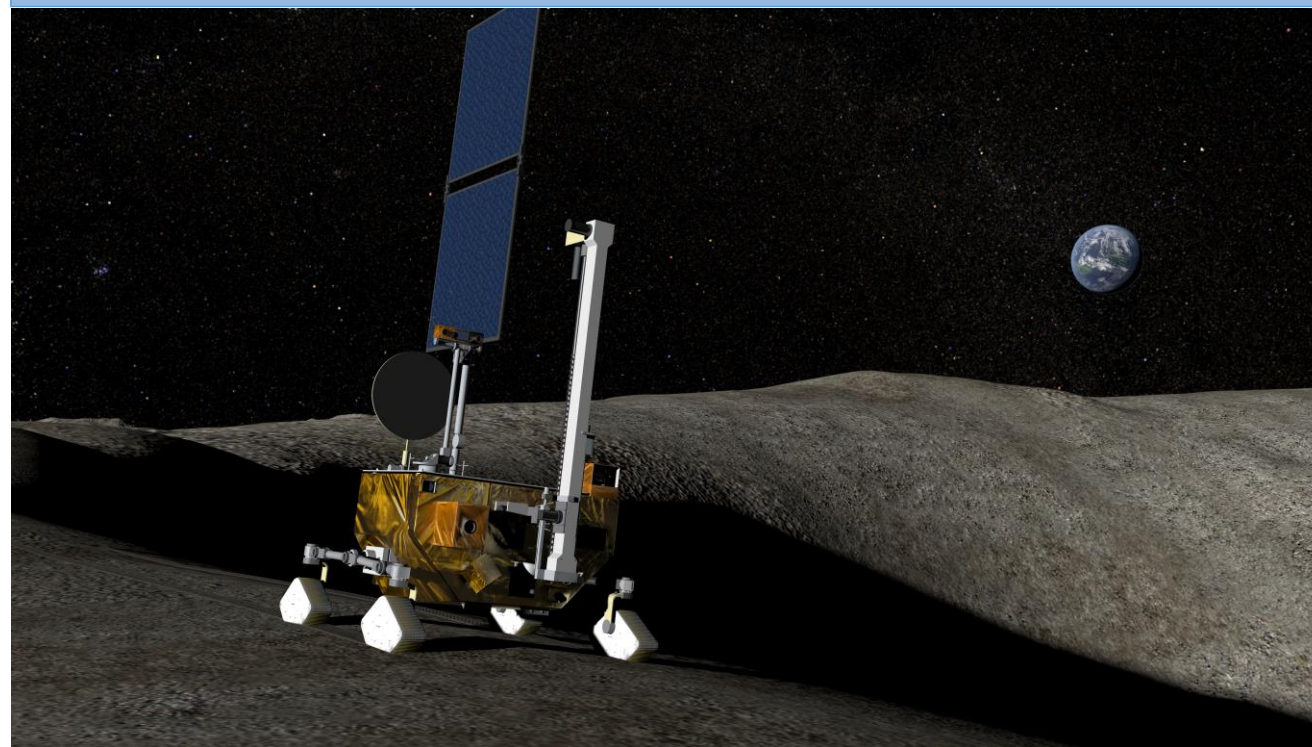
Smart Lander for Investigating Moon (SLIM)

- 100m precision landing tech demo
- Mass: 210kg (Wet) / 190kg (Dry)
- To be launched as early as 2023
- Landing Site: Vicinity of “Sea of Nectar”



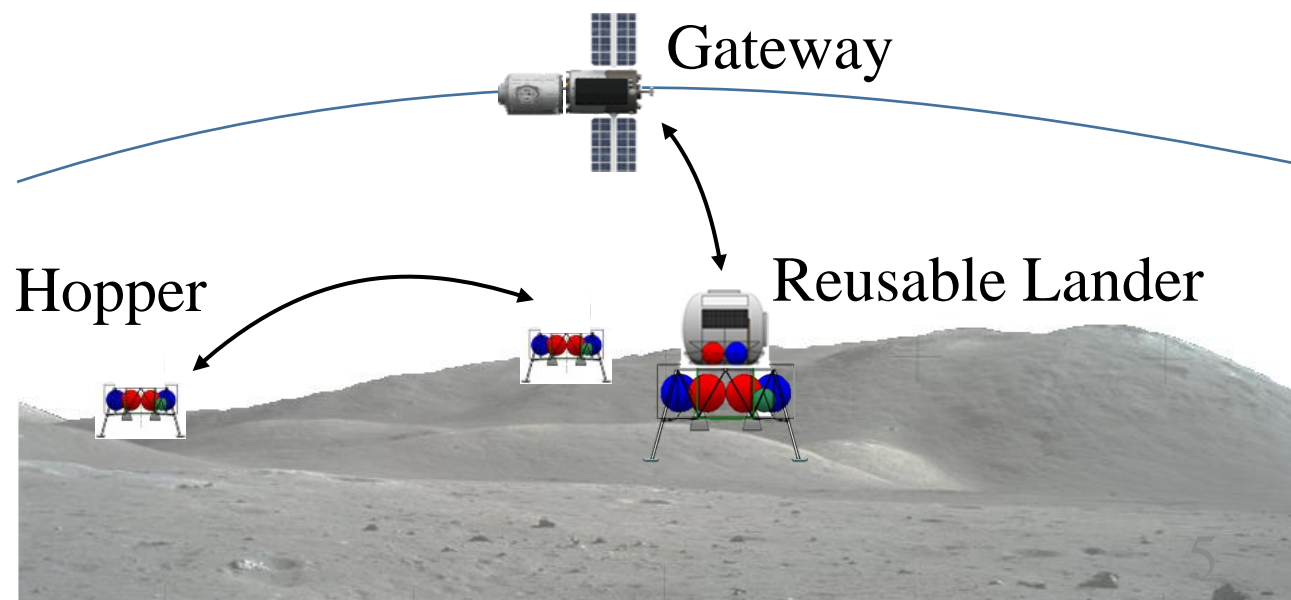
Lunar Polar Exploration (LUPEX)

- Explore the south pole region by 350kg class rover
- Investigate availability of water-ice resources
- Planned to be launched in 2025
- Collaborated w/ Indian Space Research Organisation



JAXA's vision for ISRU on the Moon

Ensure long-term sustainability of lunar surface exploration by in-situ production of O_2 and H_2 from regolith with water contents to refill reusable landers and hoppers.

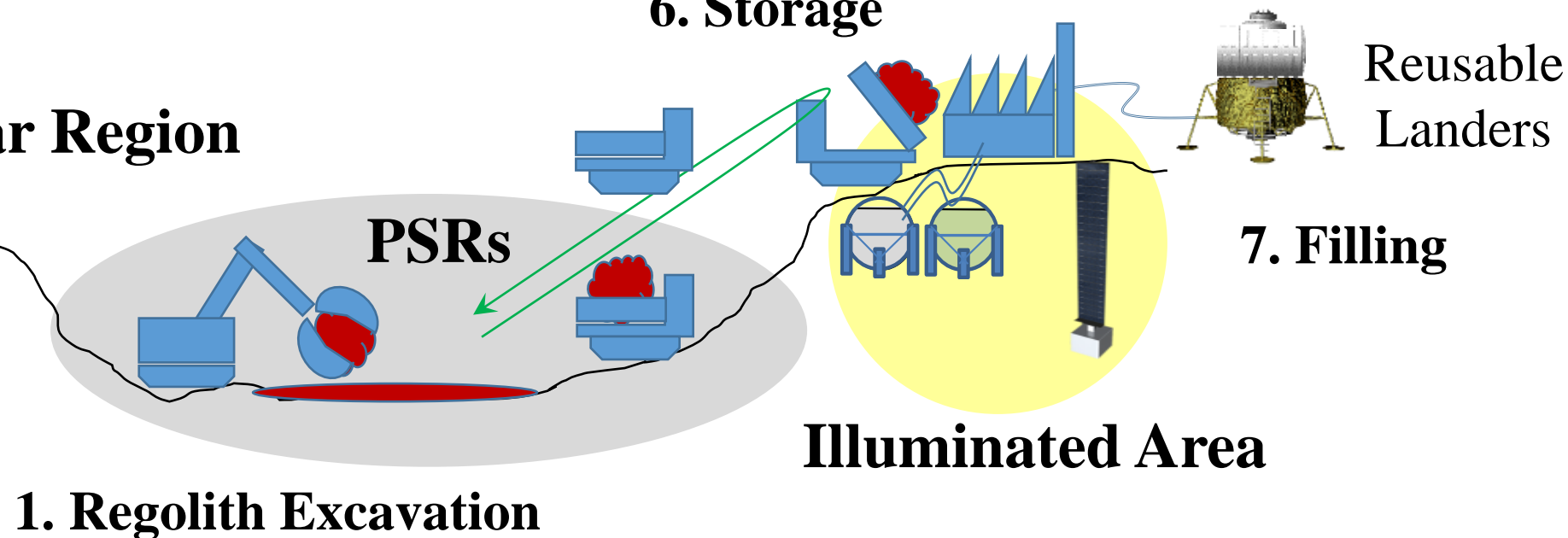


2.1 Concept

Lunar ISRU Plant

2. Water Extraction
3. Condensation & Purification
4. Electrolysis
5. GOX/GH₂ Liquefaction
6. Storage

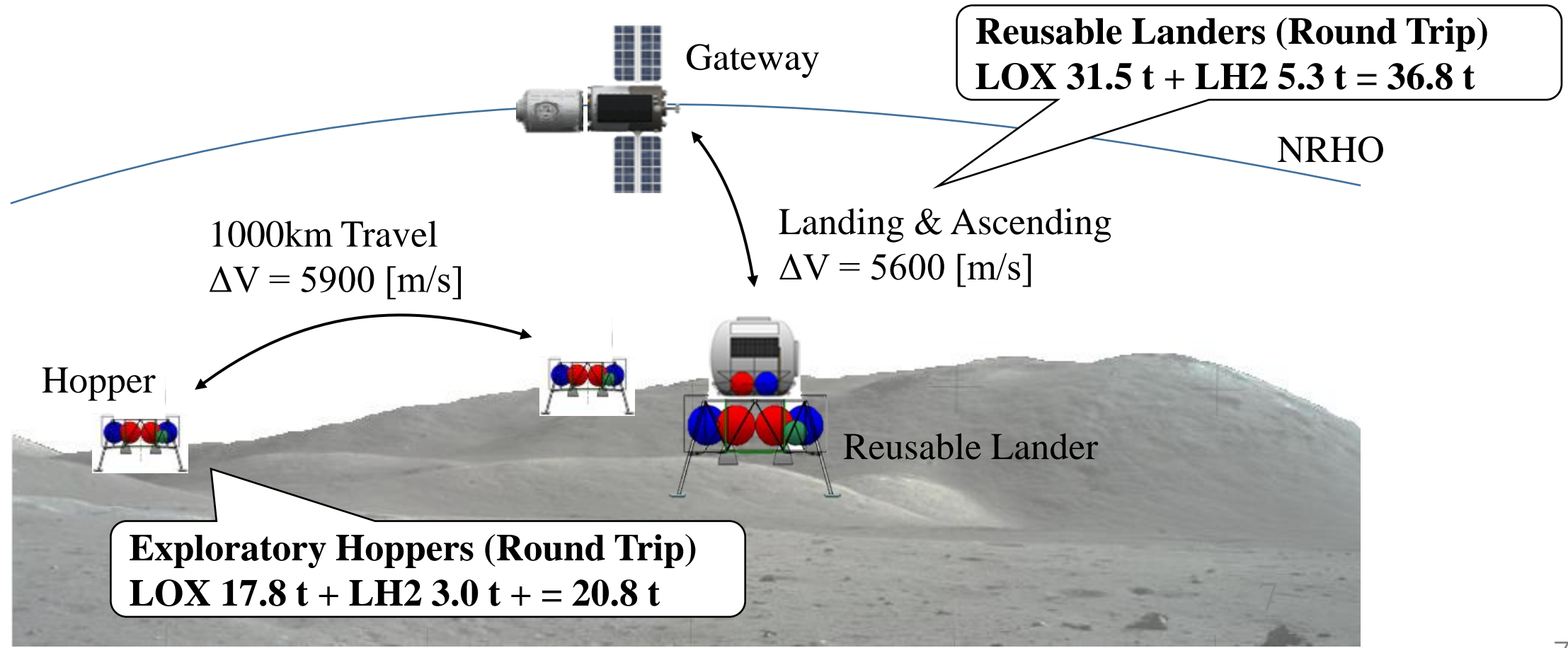
Lunar Polar Region



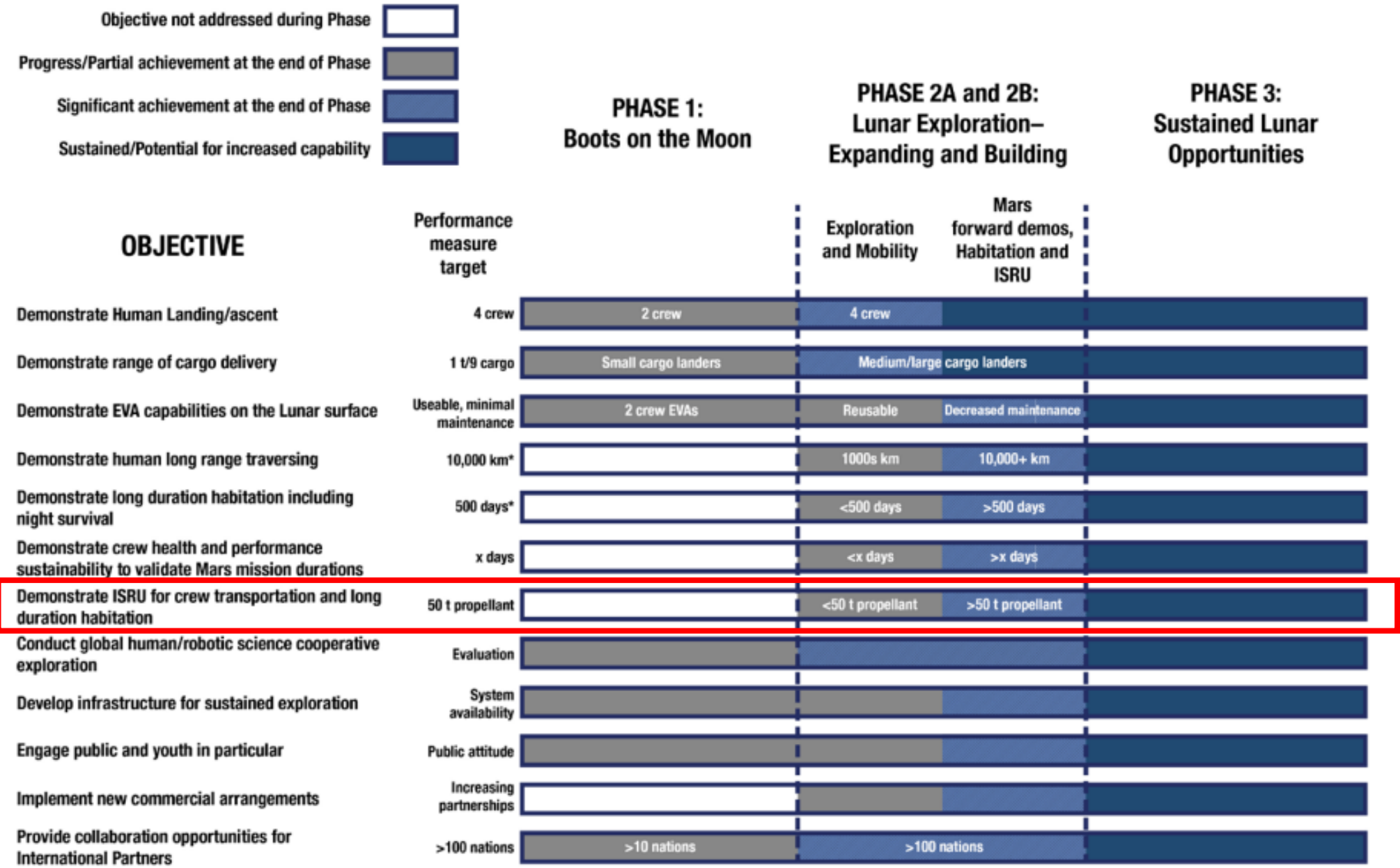
2.2 Production Rate



Production rate of propellant: 57.6 ton/year
(LOX 49.3 ton/year, LH2 8.3 ton/year)



2. Baseline requirements for a lunar ISRU plant



3.1. Overview



JAXA carried out conceptual study of a lunar ISRU Plant with JGC Corporation.

- ◆ One of the largest total engineering companies in the world.
- ◆ Headquartered in Yokohama, Japan.

3.1. Overview

Qualitative & quantitative studies of a plant system to meet baseline requirements.

- ◇ Block Flow Diagram
- ◇ Material Balance
- ◇ Key Technical Elements
- ◇ Mass & Size Estimation
- ◇ Site Selection



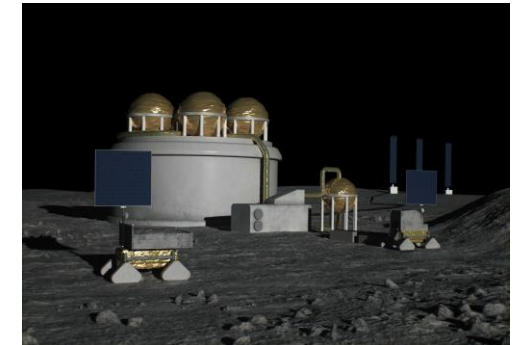
Discussions on technical & managerial issues.

- ◇ Plant System
- ◇ Subsystem
- ◇ Construction & Infrastructure
- ◇ Operation
- ◇ Project Management



Formulation of R&D strategy and project plans.

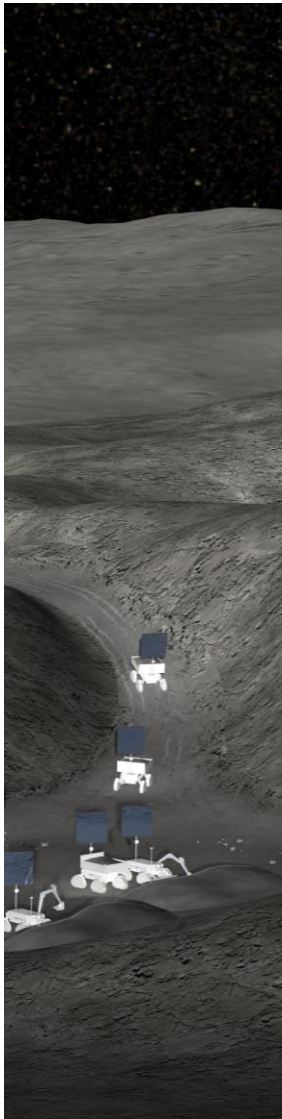
- ◇ Pilot Plant
- ◇ Full-scale Plant



Overview of conceptual study of a lunar ISRU plant

3.2. Key Technical Elements

Process	Key Technical Elements on Processing (Examples)
Regolith Excavation	<ul style="list-style-type: none"> • Low-power regolith excavation • Remote control and automatic operation of construction machinery • Decentralized control of multiple construction machinery
Water Extraction	<ul style="list-style-type: none"> • Low-power water extraction (extraction at a plant vs on-site) • Efficient heating (magnetron vs electric heating vs solar power) • Thermal insulation of heating container • Dust mitigation & Effect analysis of regolith contamination • Treatment of residue after water extraction • Water vapor adsorption on inner walls of pipelines
Condensation & Purification	<ul style="list-style-type: none"> • Efficient condensation after water extraction • Design of water tank in consideration of 1/6G



3.2. Key Technical Elements

Process	Key Technical Elements on Processing (Examples)
Electrolysis	<ul style="list-style-type: none"> • Efficient electrolyzing methods in 1/6 gravity • Reduction in weight of electrolysis • Dehumidification of GOX/GH2 prior to liquefaction
Liquefaction	<ul style="list-style-type: none"> • Energy-efficient cooling mechanism (Vapor-compression refrigeration vs Magnetic refrigeration) e.g. <u>A</u>ctive <u>M</u>agnetic <u>R</u>egenerative <u>R</u>efrigeration: AMRR
Storage	<ul style="list-style-type: none"> • Lightweight cryogenic storage tanks with material compatibility • Boil-off reduction • Energy-efficient recondensation method
Propellant Filling	<ul style="list-style-type: none"> • Boil-off reduction during propellant filling operation • Remote control and automatic refueling



3.3. Mass & Size Estimation

❑ Mass of a whole plant system: 293 ton

(*) Preconditions

Production rate: 57.6 ton/year

Site: Prospective construction site in south pole region

Distance between power facility to a ISRU plant: 20 km

Solar illumination: Ave. 50 %

Power Generation: Solar Photovoltaics (PV)

Power Transmission: Power Cables (Al conductor cables)

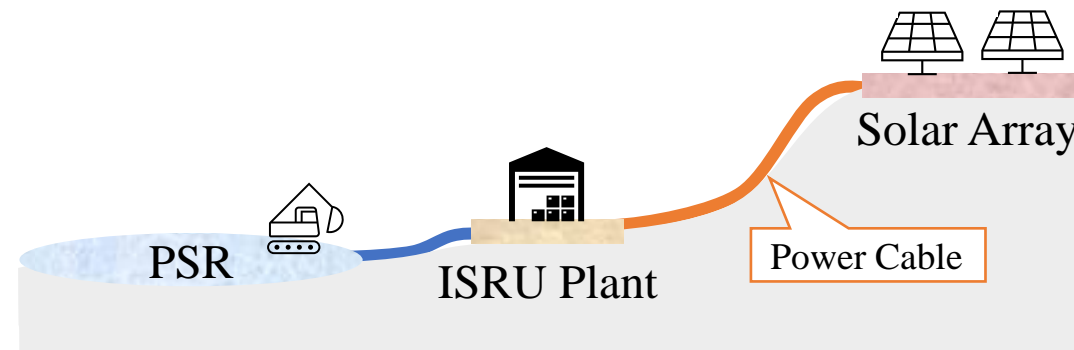
Power Storage: Fuel Cell

❑ Total area of Photovoltaics: approx. 2,000 m²

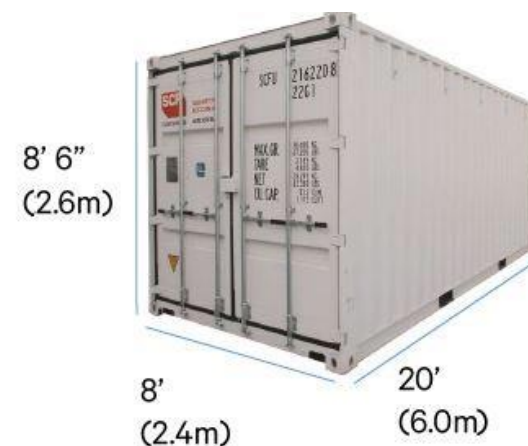
❑ Target volume of a ISRU plant: < 33.1m³ (= 1,169 cu ft.)

≡ Inside cubic capacity of 20ft. ISO Container

❑ Total area of heat rejection radiators: approx. 300 m²

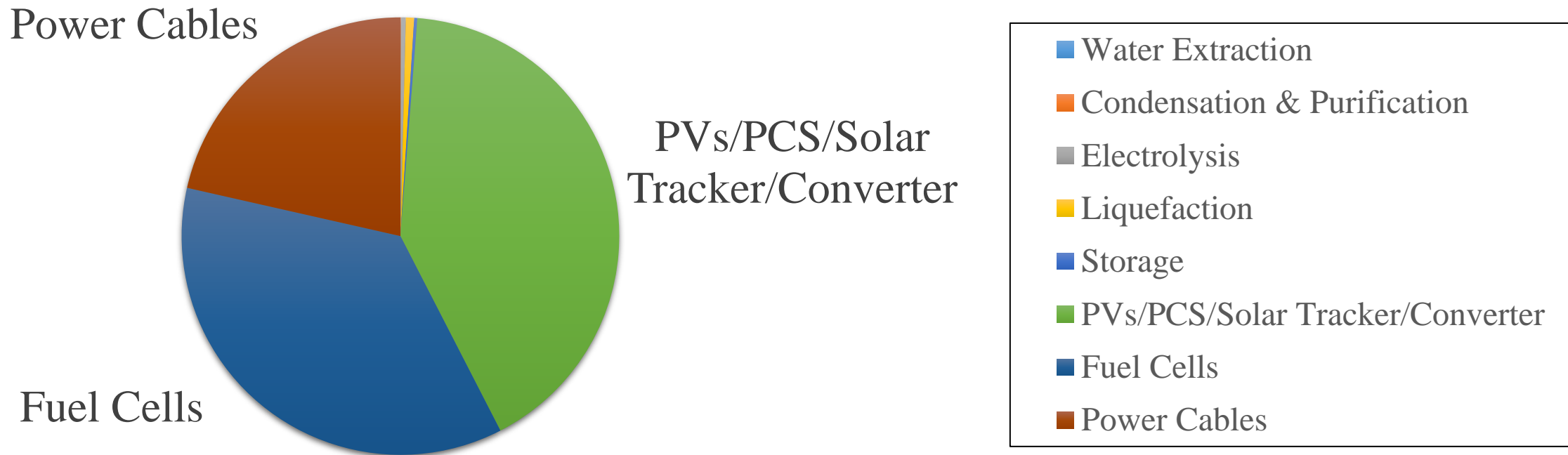


Configuration of a lunar ISRU Plant



20ft. ISO Container

3.3. Mass & Size Estimation



Drastic weight reduction of power elements (PVs / power conditioning system (PCS) / solar tracker / converter, fuel cells, power cables) and heat rejection radiators is essential given that over 95% of the total mass comes from those elements according to our estimation.

3.4. Construction Method

- ❑ Modular construction method, which is developed method of LNG plant construction with the aim of speedy construction, is applicable to a lunar ISRU plant.



Module Rifting



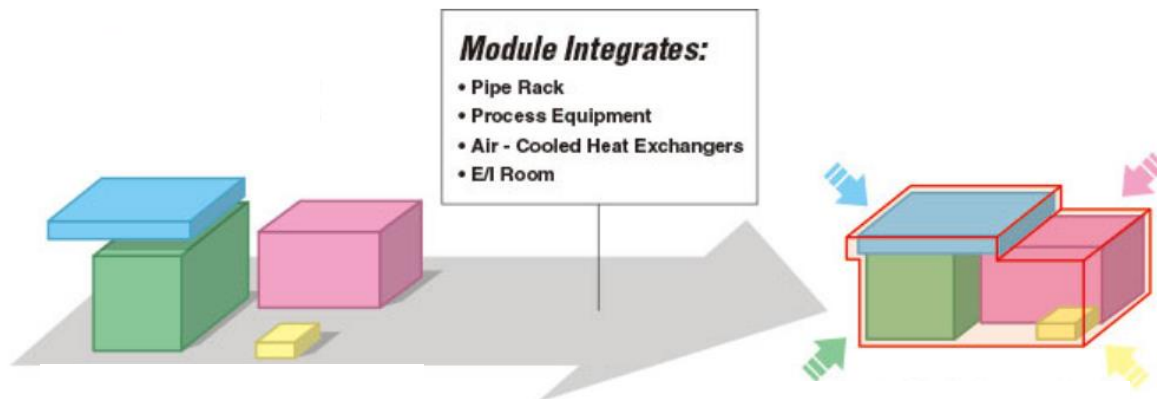
Maritime Transportation



Delivery to Construction Site



Land Transportation



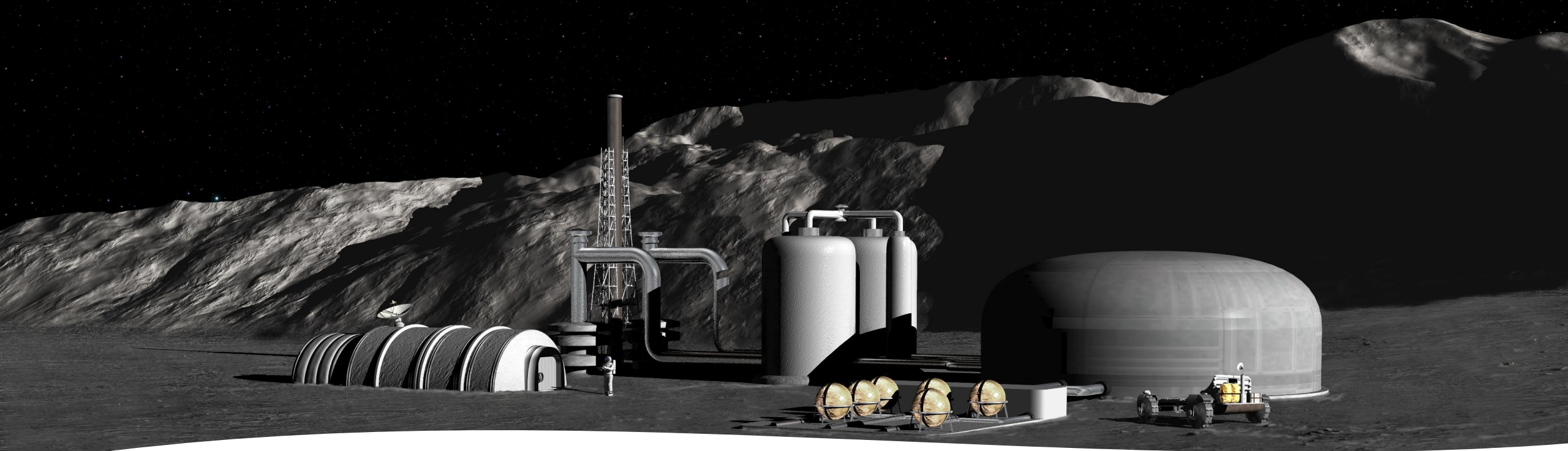
SPMT (Self-Propelled Modular Transporter) / Mammoet

4.1. Requirements Definition & Flowdown

- Define system requirements of a lunar ISRU plant taking account of stakeholders' expectations and constraints
- Flow down system requirements from a plant system level to a subsystem level.

4.2. Building an R&D Strategy and Project Plans

- Concept and technology development from a system integration standpoint.
- Deepen fundamental research for key technical elements based on requirements.
- Research on construction & infrastructure technology on a lunar surface in cooperation with general contractors and construction machinery manufacturers.
- Demonstrate LOX/LH2 production technologies from regolith simulant with water contents on the ground through development of an integrated ISRU system in the early 2020s.
- Develop a pilot plant for a sub-scale tech demo and construct a large-scale plant on the Moon.



Pilot Plant

- **Perform a sub-scale tech demo on the Moon in the late 2020s - early 2030s.**
- **Produce potable water (340 kg/year TBD) and/or breathable O₂ (150 kg/year TBD) from regolith.**

Large-scale Plant

- **Commence to build a lunar ISRU plant in the late 2030s and start full-scale operation by 2040.**
- **Produce LOX (49.3 ton/year TBD) and LH₂ (8.3 ton/year TBD) from regolith to refill spacecrafts.**

NOTICE: Above-mentioned plans and requirements stem from the conceptual study of a lunar ISRU plant by JAXA and JGC Cooperation. Target production rate may change. Not budgeted by the Government of Japan at this moment.



“The future is not a gift: it is an achievement.
Every generation helps make its own future.
This is the essential challenge of the present.”

— Robert Kennedy —

